

Experimental Behavior of Iron-based Shape Memory Alloys under Cyclic Loading Histories

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Abstract

The present paper investigates the behavior of iron-based shape memory alloys (Fe-SMAs) subjected to cyclic inelastic straining by means of uniaxial coupon experiments. The tests feature round bar coupons subjected to a broad range of uniaxial cyclic strain histories representative of earthquake loading. The experimental results suggest that the Fe-SMA under investigation exhibits an asymmetric stress-strain relation, with limited superelastic behavior. It was found that the post-yield/phase transformation behavior of the Fe-SMA alloy is both strain-rate and temperature-dependent. Quantitative comparisons with structural steels subjected to nominally identical cyclic strain histories indicate that, although the studied Fe-SMA has a similar energy dissipation per loading excursion with respect to conventional S355J2+N, the Fe-SMA's hardening response is appreciably higher, leading to comparatively larger elastic strain energies being stored.

Keywords: Iron-based shape memory alloys, Inelastic cyclic loading, Temperature dependency, Earthquake loading, Superelasticity, Smart materials

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20 1. Introduction

21 Shape memory alloys (SMAs) are advanced materials exhibiting superelastic be-
22 havior and shape memory effect (SME) under thermomechanical loading. These two
23 properties are due to a reversible phase transformation between martensite and austen-
24 ite. Prior studies in the context of earthquake engineering suggest that the superelas-
25 tic behavior of Nickel-Titanium SMAs (NiTi-SMAs) is promising for providing both
26 energy dissipation and re-centering capabilities [1]. Examples in large-scale civil engi-
27 neering applications using NiTi-SMAs include self-centering buckling-restrained braces
28 [2], dampers with self-centering capabilities [3], as well as self-centering beam-column
29 connections [4]. An extensive summary on the wide use of SMAs in seismic and wind
30 engineering with emphasis on buildings, bridges [5, 6, 7, 8] as well as coastal structures
31 [9] can be found in literature. In prior work, the mechanical behavior of NiTi SMAs
32 has been characterised under cyclic tension and torsion for potential use in seismic ap-
33 plications [10, 11]. The main aspects that were investigated were the re-centering and
34 energy dissipation capabilities when considered as energy dissipation elements. Because
35 these applications were limited to wires the uniaxial compressive stress-strain response
36 was not characterized. Similarly, Padgett et al. [12] has proposed and validated exper-
37 imentally an SMA restrainer cable for seismic risk mitigation of bridges. Fugazza et al.
38 [13] further explored the mechanical properties of SMAs when employed in devices for
39 vibration control of buildings along with uniaxial constitutive modelling for nonlinear
40 response history analysis. In a recently published study, Vignoli et al. [14] explored
41 the dynamics of seismically resistant structures equipped with SMA composites and
42 demonstrated that they can be an effective alternative for seismic risk mitigation as
43 opposed to other response modification devices

44 However, NiTi-SMAs are prohibitively expensive for widespread use in large-scale
45 infrastructure projects. Alternatively, iron-based (Fe-)SMAs are increasingly being
46 produced at a lower cost than Ni-Ti-SMAs.

47 To date, the mechanical behavior of a broad range of Fe-SMA alloys with different

48 chemical compositions has been explored. Our current paper focuses on a refined
49 version of an Fe-SMA developed by Janke [15], with the composition Fe-17Mn-5Si-
50 10Cr-4Ni-1(V,C) (ma,-%). This material has been, and is currently, used for the pre-
51 stress strengthening of civil infrastructure, with notable emphasis on existing bridges
52 [16, 17, 18]. A general review of Fe-SMA properties relevant for retrofitting existing
53 infrastructure under long term loading is provided in Cladera et al. [19]. Markedly, Fe-
54 SMAs exhibit a lower level of superelasticity compared to NiTi-SMAs, however, they
55 still show an appreciable shape memory effect [19].

56 The austenite/martensite phase composition of Fe-SMAs, and its evolution under
57 thermomechanical loading, is particularly relevant since it governs the superelastic and
58 SME properties of the material. Lee et al. [20] studied the phase evolution under
59 tensile thermomechanical loading representative of the one used to pre-stress rein-
60 forced concrete structural components suitable for bridge applications. Lee et al. [20]
61 identified three distinct phases when the material is loaded under monotonic tensile
62 loading: 1) γ' -austenite, 2) ε -martensite, and 3) α' -martensite. The α' -martensite is an
63 irreversible martensite variant that cannot be transformed to austenite, thereby pre-
64 venting the SME. Moreover, Lee et al. [20] established the bounds on the temperature
65 and stress at which martensite-to-austenite (reverse) phase transformation may occur.
66 The above studies reveal that the reverse transformation of the Fe-SMA can occur at
67 room temperatures below tensile stress levels of around 100 MPa. The α' -martensite
68 phase was shown to only appear at temperatures above 100 °C, and therefore it does
69 not seem to influence the Fe-SMA's behavior at room-temperatures under monotonic
70 tensile loading. However, there are no bounds identified or established in the case that
71 the material experiences compressive stresses as may be expected in the dissipative
72 elements of steel structures in seismic regions.

73 The general consensus from an extensive literature review on the material properties
74 of Fe-SMAs is that there are currently no comprehensive studies to characterize their
75 behavior under cyclic inelastic straining similar to that seen during earthquake loading.
76 Typical mechanical properties of the Fe-SMA under study such as the yield stress, the

77 elastic modulus, the ultimate tensile stress, as well as some low-cycle fatigue response
78 features, have been identified in prior works [21, 22, 23]. Notable in Ghafoori et al.
79 [22] are experimental results that suggest that strain-rate effects are present in the
80 Fe-SMA's behavior. Therein, an increase in yield stress of ≈ 33 MPa between a test
81 carried out at ≈ 0.03 %/sec and a test carried out at ≈ 1.4 %/sec is reported. In another
82 study, Hosseini et al. [23] found that the maximum tensile stress of the Fe-SMA strips
83 was ≈ 1000 MPa and they could elongate up to ≈ 55 % prior to fracture. Furthermore,
84 although Ghafoori et al. [24] have studied the elevated temperature behavior of Fe-
85 SMA alloys (under a fire exposure) up to 1000°C , the self-heating capability of the
86 alloy under different strain rates has not been studied so far.

87 Although the above findings are of intrinsic value, material properties have hitherto
88 been assessed by imposed loading histories that are deemed to be inadequate for seismic
89 loading. Tests to date have mainly focused on characterizing mechanical and SME
90 behavior solely under monotonic tensile or cyclic tensile loading, underscoring a lack of
91 knowledge of this Fe-SMA under compressive strain demands that are likely to occur
92 during earthquake loading. Other overlooked factors include the influence of strain
93 level, strain rate, and temperature dependency on the material's SME or superelasticity
94 (when present) at loading rates representing earthquake loading. Here, typical testing
95 strain rates are generally not consistent with those expected during an earthquake
96 event [25].

97 On the demand side, uncertainties (often titled record-to-record variability) in the
98 earthquake ground motion hazard, in terms of both amplitude and frequency content,
99 typically cause variability in the response characteristics of new and existing structures.
100 Consequently, demands on the material itself can be highly variable and a need exists
101 to assess its response under distinct loading histories. Studies with the objective of
102 characterizing steel material behavior for seismic applications have addressed the issue
103 of uncertainty by defining uniaxial strain demand protocols derived from the response
104 of moment frames to a rich set of ground motions [26]. The same approach is used
105 herein in an effort to comprehend the material behavior under such loading conditions.

106 The objective of the present paper is to characterize the macroscopic behavior of
107 the Fe-17Mn-5Si-10Cr-4Ni-1(V,C) (ma,- %) Fe-SMA subjected to uniaxial tensile and
108 compressive inelastic straining consistent with demands induced by earthquake loading.
109 This objective is achieved through uniaxial cyclic experiments conducted on round bar
110 coupon specimens. The next part of the paper describes the testing procedures and test
111 apparatus used to perform the experiments on Fe-SMAs. The second part presents the
112 results of the experimental campaign conducted on the Fe-SMA specimens. The third
113 part contrasts experimental findings with available data from Fe-SMA materials with
114 nominally identical chemical compositions that have been reported in the literature, as
115 well as with results from commonly used structural steels in earthquake and structural
116 engineering.

117 **2. Methodology**

118 The methodology to characterize the hysteretic behavior of the Fe-SMA comprises
119 of an experimental campaign based on the ASTM E8/E8M guidelines [27] adapted for
120 cyclic loading. This section discusses the details of the test apparatus along with the
121 basic geometries of the Fe-SMA specimens (see Section 2.1), the employed strain-based
122 load protocols (see Section 2.2), as well as the key analysis metrics used in interpreting
123 the results and their data processing (see Section 2.3).

124 *2.1. Test specimens and experimental apparatus*

125 The geometry of a typical test specimen is presented in Figure 1. Particularly, the
126 specimens are smooth round bar coupons designed to delay buckling until high com-
127 pressive strains (i.e., larger than 5 %). This is shown in Figure 2, where the specimen
128 is at 5 % inelastic uniaxial compressive strain and there is no buckling. Referring to
129 Figure 1, the unreduced diameter of the specimen is 12 mm (with M12 threading),
130 and the diameter of the reduced section is 6 mm. The gauge length of the specimen is
131 designed such that buckling is prevented under inelastic uniaxial compressive loading
132 as discussed in detail in de Castro e Sousa et al. [28]. To ensure the reliability of local

133 strain measurements with the 8 mm gauge length extensometer, we verified the same
134 measurements with a video-extensometer that was used in preliminary verification test-
135 ing of the overall test apparatus. The specimens are manufactured from coiled Fe-SMA
136 reinforcing bars (rebars). To produce the specimens, the rebar coils are straightened,
137 then the round bar coupons are machined according to the specifications in Figure 1.

138 The tests are conducted using a Schenk self-reacting frame available at EPFL Struc-
139 tural Laboratory (GIS), shown in Figure 3a, where the cross-beam is controlled by an
140 electric actuator. A cross-section and plan-view of the test apparatus (Figures 4a
141 and 4b, respectively) aid to explain the mounting process of a test specimen. In brief,
142 the upper and lower adapter plates are bolted and prestressed to the upper cross-beam
143 and the base, respectively. The three-piece outer ring, shown in Figure 4b, enables ten-
144 sile loading of the specimen, while compressive loads are transferred directly from the
145 counter nut to the lower plate. Importantly, the straightness of the mounted specimen
146 is ensured by the straightness of the specimen itself and the perpendicularity of the
147 machined threaded hole of the top plate. The lower plate is adjusted until the lower
148 counter nut on the specimen enters its socket and is in contact with the plate. The
149 nut and counter are then tightened against each other. Afterwards, the lower plate is
150 bolted and pre-stressed to mobilize friction. The socket also provides lateral support,
151 so that lateral forces can be absorbed by the lower plate and transferred to the reacting
152 frame.

153 Figure 3b shows the test apparatus with a specimen installed in place along with
154 the measurement devices. In particular, local strain measurements are obtained using
155 an MTS extensometer with an 8 mm gauge length, force measurements are taken
156 directly from the load cell attached to the top cross-beam of the Schenk frame, and
157 the cross-beam displacement is recorded with a linear potentiometer. The surface
158 temperature of a test specimen is recorded using a PT100 temperature sensor. Two
159 cameras (shown in Figure 3a) are used to record videos at 60 frames per second in
160 order to visually detect buckling of the specimen under compressive loading. This
161 measure is put in place since visually detectable buckling deformations are perceptible

much sooner than degradation in the load-displacement curve. Only results prior to specimen buckling are considered to be meaningful at the material scale. Otherwise, the geometric nonlinearities are conflated with the sole material response.

The tests are controlled using a Proportional-Integral-Derivative (PID) algorithm with a Walter+Bai PCS8000 controller. All tests are conducted in load control up to $0.65f_y$, where f_y denotes the nominal yield stress of the Fe-SMA assumed to be ≈ 450 MPa [22]. The test execution is then switched to strain control using the MTS extensometer. Strain control allows for precise control over the desired strain rate depending on the load protocol of interest. Portions of the tests past 12.5 % strain are controlled using the cross-beam displacement due to the limit of the MTS extensometer. Stable control of the test becomes a critical issue when high strain rates are used, i.e., when $\dot{\epsilon} \geq 1$ %/sec. For this reason, the value of the proportional gain is reduced from P (the best-fit gain calibrated to a square waveform) to $\tilde{P} = 1/3P$ to compensate the decrease in the tangent modulus of the specimen. The integral parameter (I) is left the same and the derivative (D) is set to zero (no compensation from the time change of the signal). It was found that this approximation allows for stable control of the testing machine at strain rates up to 8 %/sec.

2.2. Load protocols

The load protocols (LPs) comprise uniaxial strain histories applied to the Fe-SMA coupons. These are based on prior work by Suzuki [26], de Castro e Sousa et al. [28]. In brief, these protocols have been established based on nonlinear response history analyses of steel moment resisting frames featuring columns made of conventional or high performance steel. The suite of LPs features: a uniaxial monotonic test (LP1); two tests (LP2, LP3) representing different ground motion frequency characteristics followed by a large monotonic tensile push till specimen fracture; two constant amplitude tests (LP4, LP5) representative of long-duration ground motions that cycle a structure symmetrically at modest inelastic lateral drift demands (i.e., mean effects tend to be zero); three incremental tests (LP6, LP7, LP8) representative of ordinary earthquake

190 records scaled at design basis seismic intensities (i.e., 10 % probability of exceedance
191 over 50 years) that cycle a structure symmetrically; and a test termed "random" (LP9)
192 that is representative of ordinary, far-field and near fault ground motions scaled at a
193 seismic intensity associated with a probability of exceedance of 2 % in 50 years.

194 The selected strain rates used in all the aforementioned protocols are informed
195 by international material testing standards [27, 29]. The strain rate of 0.03 %/sec is
196 selected for strain demands less than 2 % to load the specimens quasi-statically prior-
197 to, and during, initial yielding, whereas the rate of 0.8 %/sec is selected for strain
198 demands past this point.

199 The 2 % constant strain amplitude test (LP5) is chosen to study strain rate effects
200 on the Fe-SMA's hysteretic behavior. Particularly, we conduct tests with the two
201 standard strain rates (0.03 %/sec and 0.8 %/sec), complemented by a test at the rate
202 of 8 %/sec. While the former rates are representative of quasi-static loading, the later
203 approximates typical expected strain rates in steel structures during seismic events [25].
204 All tests are carried out at a room temperature of around 22 °C. Consistency of the
205 experimental results between specimens subjected to LP5 is evaluated by conducting
206 two additional tests per strain rate. Vis-a-vis the above discussion, the experimental
207 program comprises 14 specimens in total.

208 In brief, the following tests are conducted:

- 209 • LP1: consists of monotonic tensile excursion until specimen fracture,
- 210 • LP2: consists of one cycle at 1 % strain amplitude in tension and compression,
211 followed by a tensile excursion until specimen fracture,
- 212 • LP3: consists of one cycle at 4% strain amplitude in tension and compression,
213 followed by a tensile excursion until specimen fracture,
- 214 • LP4: consists of 50 cycles with a constant strain of 1 % in tension and compres-
215 sion, with a strain rate of 0.03 %/sec,

- 216 • LP5_SR: consists of 50 cycles with a constant strain of 2 % in tension and com-
217 pression, with a strain rate of 0.03 %/sec,
- 218 • LP5_MR: consists of 50 cycles with a constant strain of 2 % in tension and
219 compression, with a strain rate of 0.8 %/sec,
- 220 • LP5_FR: consists of 10 cycles with a constant strain of 2 % in tension and com-
221 pression, with a strain rate of 8 %/sec to ensure stability of the test for high strain
222 rates. The specimen is then left for 5 minutes to cool down. In turn, 50 more
223 loading cycles are imposed with identical strain amplitude and rate. Therefore,
224 a total of 60 cycles was performed for LP5_FR,
- 225 • LP6: consists of incrementally increasing strain amplitude of 0.5 % until 8 %
226 strain with a strain rate of 0.03 %/sec until 2 % strain and 0.8 %/sec after,
- 227 • LP7: consists of incrementally increasing strain amplitude of 1 % until 8 % strain
228 with a strain rate of 0.03 %/sec until 2 % strain and 0.8 %/sec after,
- 229 • LP8: consists of incrementally increasing strain amplitude of 2 % until 8 % strain
230 with a strain rate of 0.03 %/sec until 2 % strain and 0.8 %/sec after,
- 231 • LP9: consists of a “random” strain amplitude protocol. Due to an error in the
232 test control, LP9 used herein is different than the standard “random” protocol
233 discussed in Suzuki [26] and de Castro e Sousa et al. [28].

234 Results past compressive strains concurrent with buckling are not considered to be
235 meaningful in the context of the present study and they are disregarded. A temperature
236 gauge was used in most of the tests to track the variations in temperature while loading
237 excluding LP1 and LP6.

238 2.3. Key metrics and data processing

239 The macroscopic material behavior of the Fe-SMA determined in this paper is based
 240 on true stress-true strain values. Engineering strain is computed using Equation 1,

$$241 \quad \varepsilon_{eng} = \frac{L - L_0}{L_0} \quad (1)$$

242 where $L - L_0$ is the displacement measured from the extensometer, and $L_0 = 8$ mm
 243 from the extensometer gauge length. True strain is then calculated using Equation 2,

$$244 \quad \varepsilon = \ln(1 + \varepsilon_{eng}) \quad (2)$$

245 Engineering stress is computed using Equation 3,

$$246 \quad \sigma_{eng} = \frac{F}{A_0} \quad (3)$$

247 where F is the force measured from the load cell of the test apparatus, and A_0 is the
 248 initial cross sectional area of a specimen, computed using a diameter as measured by
 249 a conventional caliper gauge. True stress is then calculated using Equation 4,

$$250 \quad \sigma = \sigma_{eng}(1 + \varepsilon_{eng}) \quad (4)$$

251 When subjected to inelastic strains, Fe-SMAs can develop both plastic and trans-
 252 formation strains [19]. Only the total strain is measured in the experiments, therefore,
 253 it is not possible to differentiate between the plastic and transformation strains. Hence,
 254 the term compound strain, referring to the combination of plastic and transformation
 255 strain is introduced to describe the Fe-SMA material behavior.

256 The elastic modulus E is determined with a line of best fit adjusted to the true
 257 stress-true strain data up to a stress of 100 MPa in the first loading excursion. The
 258 0.2 % offset yield stress $\sigma_{p,0.2\%}$ is calculated according to ASTM [27]. A 0.01 % offset
 259 yield stress $\sigma_{p,0.01\%}$ is obtained similarly to $\sigma_{p,0.2\%}$. The ultimate tensile stress is cal-
 260 culated for load protocols that include a monotonic tensile excursion up to specimen
 261 fracture. The extensometer was removed prior to specimen fracture (12.5 % strain is

approximately the gauge's limit) in the current experimental program and, therefore,
the ultimate tensile stress, σ_u , is defined based on the engineering stress,

$$\sigma_u = \frac{F_{max}}{A_0} \quad (5)$$

where F_{max} is the maximum force measured by the load cell. With respect to 0.2 %
offset yield stress, for the Fe-SMA we do not differentiate between plastic deformation
and phase transformation mechanisms.

A normalized parameter h is introduced in order to assess the level of cyclic hard-
ening in a constant amplitude test present in the material in a quantitative manner. In
particular, Equation 6 introduces the variable h , which corresponds to $\Delta\sigma$, that is the
increase in stress due to cyclic hardening, divided by the 0.2 % offset yield stress. The
values of $\Delta\sigma$ are computed as the increase in maximum stress due to cyclic hardening
between the first loading cycle and the last loading cycle in the same loading direction
of the constant 2 % strain amplitude with the 0.03 %/sec strain rate load protocol
(LP5_SR),

$$h = \frac{\Delta\sigma}{\sigma_{p,0.2\%}} \cdot 100 \quad [\%] \quad (6)$$

Furthermore, to determine the Fe-SMA's potential for energy dissipation through
cyclic straining, the portion of the area under the stress-strain relationship correspond-
ing to the compound strain is computed using two different imposed load histories. The
2 % constant strain amplitude and the 1 % incrementally increasing strain amplitude
protocols are selected for this purpose. Equation 7 represents the dissipated energy
normalized by the elastic strain energy calculated using the mean 0.2 % offset yield
stress and mean elastic modulus.

$$E_{cum}^d = \frac{\int \sigma(\varepsilon - \frac{\sigma}{E}) d\varepsilon}{\frac{1}{2} \frac{\sigma_{p,0.2\%}^2}{E}} \quad (7)$$

The trapezoidal integration rule is used to integrate Equation 7.

The engineering strain at fracture, ε_{frac} , is computed using the relation between the
cross-head displacement and the engineering strain. The relation is the extrapolated

for strain amplitudes greater than 12.5 %, where the MTS extensometer was removed due to its limit. The reported values reported herein for the engineering strain at fracture should only be considered as indicative. The authors do not think that the employed approach that has been used in prior studies as well is rational to extract the engineering strains at fracture, thereby leading to large reported values.

3. Results and discussion

3.1. Mechanical properties

Mechanical properties pertinent to structural analysis and design are summarized herein. These properties include the elastic modulus, E , the offset yield stresses, $\sigma_{p,0.2\%}$ and $\sigma_{p,0.01\%}$, the ultimate stress, σ_u , and the engineering fracture strain ε_{frac} . The material's mechanical properties are calculated according to the methodology described in Section 2.3, results for each parameter of interest, including the mean and coefficient of variation (CoV) based on all tests, are summarized in Table 1. The average results are: $E = 184$ GPa (CoV = 5.11 %), $\sigma_{p,0.01\%} = 267$ MPa (CoV = 22.7 %), and $\sigma_{p,0.2\%} = 450$ MPa (CoV = 13.7 %), and $\sigma_u = 950$ MPa. The LP5 medium- and fast-rate results are not included in the calculation of the mean values because of difference in loading rates.

Table 2 compares the mean of some of the aforementioned material properties obtained from the experimental campaign with indicative values found in literature for nominally identical Fe-SMA alloys [21, 22, 23]. Referring to Table 2, the values obtained from the experimental campaign are within bounds established in prior studies, thereby indicating confidence in the alloy's reproducibility of mechanical properties. The same observation holds true for the mean fracture strain reported in Table 1. Referring to Table 1, the fracture strains associated with LP2 and LP3 are smaller than that of LP1 due to the cumulative damage effect from the inelastic cycle prior to the monotonic tensile excursion to fracture. Notably, regardless of the amplitude of the inelastic cycles between LP2 and LP3, the fracture strain is fairly consistent between these two tests.

312 The CoV values summarized in Table 1 suggest that the mechanical properties of
 313 the Fe-SMA exhibit appreciable variability. Particularly, the variations in the alloy's
 314 material properties are evident as the CoV for the 0.2 % offset yield stress is nearly
 315 14 %. The fairly large CoV values for the 0.2 % offset yield stress of the Fe-SMA
 316 should be contrasted to the corresponding CoV values of mechanical properties of
 317 conventional construction steels. For instance, in S355J0 steel (i.e., nominal yield
 318 stress, $f_y = 355$ MPa) the CoV around the material's yield stress is around 7 % [30].
 319 The variations in the Fe-SMA yield stress may have been caused by residual stresses
 320 imposed in the coupon specimens due to the straightening of the rebar coils.

321 3.2. *True stress-strain hysteretic behavior*

322 This section summarizes the key characteristics of the observed true stress-strain
 323 behavior of the tested Fe-SMA. Figures 5 to 13 illustrate the true stress-true strain
 324 and temperature-true strain relationships obtained from each one of the specimens.
 325 Referring to Figures 6 to 13, it can be seen that the tested Fe-SMA does not exhibit a
 326 superelastic behavior, as is discussed in [31].

327 The results indicate an asymmetric stress-strain relation of the Fe-SMA under cyclic
 328 tensile/compressive inelastic loading. This asymmetry is evident in Figures 8a to 13a
 329 (LP4, LP5_SR, LP5_MR, LP5_FR, LP6, LP7, LP8 and LP9). These figures suggest
 330 that for strain amplitudes less than approximately 3 %, there are three distinct tangent
 331 moduli when the specimen is loaded in compression, as defined in Figure 14a. In
 332 state 1, the material experiences tensile strains while under tension. During stage 2,
 333 the material experiences tensile strain demands while under compressive loading and
 334 in stage 3, the material experiences compressive strain demands while being under
 335 compressive loading.

336 The tangent moduli appear to be dependent the magnitude of accumulated com-
 337 pound strain. Referring to Figures 10 to 12a (LP6, LP7 and LP8), the three tangent
 338 moduli are present for the cycles up to 3 % strain amplitude, while they are not visible
 339 in loading cycles with strain amplitudes larger than 3 %. Furthermore, constant am-

340 plitude tests show that after a certain number of cycles this phenomenon is no longer
341 present; *c.f.* Figure 14 and the differences between its first and 50th cycle.

342 A possible reason for the observed asymmetry in the stress-strain response of the
343 material under uniaxial cyclic loading may be due to activation of the reverse trans-
344 formation at room temperatures depending on the stress level when the material is
345 loaded in uniaxial compression. This same asymmetry is also observed in NiTi SMA
346 bars [32]. Liu et al. [32] conducted uniaxial tensile and compressive tests, as well as
347 uniaxial tension/compression tests on NiTi bars. They noticed that the behavior of
348 NiTi SMA was asymmetric between tensile and compressive loading. This asymmetry
349 in the behaviour of NiTi SMA was related to different deformation mechanisms. The
350 existence of the three aforementioned distinct tangent moduli requires further studies
351 of the microstructural behavior under cyclic loading to comprehend its physical basis
352 and potentially refine the material for potential future use in seismic applications.

353 *3.3. Influence of strain rate and temperature*

354 This section discusses the influence of strain rate and temperature on the observed
355 material behavior. The influence of strain rate and temperature is particularly relevant
356 for seismic loading, which involves high strain rates up to around 8 %/sec. Not only
357 does the high strain rate itself influence material response, but it also provides a rapid
358 increase in temperatures due to yielding/ phase transformation. Depending on the
359 geometry of the component loading rate, it might not have adequate time to cool down
360 thereby accumulating heat by increasingly high temperature levels.

361 There is a marginal influence of the strain rate on the true stress-strain behav-
362 ior of the Fe-SMA material between rates of 0.03 to 0.8 %/sec. Referring to Fig-
363 ures 6a, 7a, 10, 11a, 12a and 13a (LP2, LP3, LP6, LP7, LP8 and LP9, respectively),
364 there is only a 35 MPa increase in stress when the strain rate increases from 0.03 %/sec
365 to 0.8 %/sec at a 2 % uniaxial strain amplitude. In all these tests, the temperature at
366 the strain rate transition point (first excursion past 2 % strain amplitude) is around
367 24 °C and has not increased while there is an increase in stress. Notably, the influence

of the strain rate on the resultant stress in this case is similar to that seen in mild steels [25, 33], and has been previously reported for the Fe-SMA by Ghafoori et al. [22].

Figure 9 shows the influence of the strain rate on the temperature rise. The surface temperatures in Figures 9b–9f correspond to three different strain rates. Figure 9f shows that when a strain rate of 8 %/sec (i.e., real time) is employed, the temperature measured at the surface of the gauge length increases by approximately 75 °C, while Figure 9b shows that when LP5 is conducted with a strain rate of 0.03 %/sec, the temperature measured at the same location only increases by about 2 °C. In the former, the measured temperature increases because there is much less time for the heat to be dissipated through both conduction and convection throughout the loading history. This is an important observation since in dampers dissipating energy through material yielding (e.g. buckling-restrained braces [34, 35]) the yielding core is typically wrapped with a debonding material, which in turn is encased in mortar. Therefore, the heat transfer mechanism through conduction is expected to occur at slower rates. This issue should be carefully examined in future studies associated with the use of Fe-SMAs in yield devices.

The cyclic hardening/softening behavior of the studied Fe-SMA might be dependent on the strain rate and/or the temperature. Referring to Figures 9a, 9c and 9e, for both LP5_SR and LP5_MR, cyclic hardening in tension and compression can be observed. On the other hand, for LP5_FR, cyclic softening in tension and compression is observed in the material’s response. Further experiments that isolate the effects of strain rate and temperature are required to determine the cause of the switch from cyclic hardening to cyclic softening behavior.

3.4. Comparison with structural steels

This section compares the measured material properties of the studied Fe-SMA with two other structural steels. Particularly, the S355J2+N steel (nominal yield stress, $f_y = 355$ MPa) and the S690QL high strength steel (nominal yield stress, $f_y = 690$ MPa). The former steel is typical in seismic design of conventional and high performance steel

396 structures in Europe [36], with an impact test of 27 Joules at -20°C . The latter steel is
397 a high-grade structural steel, which is heat treated through quenching and tempering
398 to improve its brittle properties. This structural steel has a low notch toughness testing
399 temperature (i.e., typically 27 Joules at -40°C).

400 In order to compare the material properties of the Fe-SMA with structural steels,
401 data for S355J2+N and S690QL steels are taken from Grigoriou and Lignos [37], Forni
402 et al. [38], Ho et al. [39]. The specimens in Grigoriou and Lignos [37] were round
403 bar coupons similar to the ones tested in the present study. The round bar coupons
404 were extracted from 25 mm thick steel plates. Whereas, the specimens in Forni et al.
405 [38], Ho et al. [39] were round bar coupons and standard rectangular tensile coupons,
406 respectively. The yield stress and elastic modulus properties of S355J2+N and S690QL
407 steels are calculated using the procedures outlined in Section 2.3 using the data from
408 Grigoriou and Lignos [37], while the values for the corresponding fracture strains are
409 taken as reported in literature [38, 39]. Forni et al. [38] determined the fracture strain of
410 S355J2+N steel by performing a monotonic tensile test with a strain rate of 0.1 %/sec
411 on a round bar coupon and computed the strain at fracture using Equation ???. Ho et al.
412 [39] determined the strain at fracture of S690QL steel using digital image correlation
413 (DIC) measurements.

414 Note that in Grigoriou and Lignos [37], tests were controlled using the cross-
415 head displacement. The strain rates at which the tests were conducted were between
416 0.01 %/sec and 0.06 %/sec. Therefore, the strain rates used in Grigoriou and Lignos
417 [37] are similar to the slow strain rate (0.03 %/sec) used in the present paper. In Grig-
418 oriou and Lignos [37], the constant 2 % strain amplitude protocol comprised 21 cycles
419 for the S355J2+N steel. Due to issues associated with the PID control, only three full
420 cycles were performed for the S690QL steel. In principle, comparisons between the two
421 structural steels and the Fe-SMA should have been established at the aforementioned
422 cycles given that the Fe-SMA was subjected to 50 constant cycles, as discussed in Sec-
423 tion 2.2. However, the difference in number of cycles is not relevant for the present
424 research, as both materials reached saturation (i.e., there is a negligible increase in the

absolute maximum value of stress between subsequent cycles) at the end of each test.

Table 3 summarizes the mean material characteristics of the Fe-SMA, S355J2+N, and S690QL steels. The mean Fe-SMA elastic modulus, 184 GPa, is less than the one commonly used for structural steels (i.e., 200 GPa). Table 3 suggests that the Fe-SMA has a higher 0.2 % offset yield stress, higher ultimate tensile stress and lower engineering fracture strain than that of the S355J2+N steel. On the other hand, the Fe-SMA has a lower yield stress, lower ultimate tensile stress and lower engineering fracture strain than those measured for the S690QL steel.

Referring to LP1 (Figure 16a), at a strain amplitude of 12.5 %, the corresponding stress is $\sigma_{12.5\%,Fe-SMA} = 931$ MPa for the Fe-SMA, $\sigma_{12.5\%,S355} = 527$ MPa for S355J2+N steels, and $\sigma_{12.5\%,S690} = 872$ MPa for the S690QL steel. These results indicate that the post-yield hardening ratio of the Fe-SMA is larger than that of both the S355J2+N and S690QL steels. The values of the parameter h recorded in Table 3 suggest that the amount of cyclic hardening of the Fe-SMA is larger than that of the S690QL steel, and equivalent to that of the S355J2+N steel. This can also be seen by the corresponding comparisons in Figure 16. The amount of hardening in the Fe-SMA material deserves much attention in future developments of this alloy for potential use in seismic applications. The reason is that the amount of hardening within a dissipative fuse (e.g., yield segment or zone) should be controlled to prevent damage in non-dissipative structural elements during strong ground motion shaking [40].

Referring to Figures 15 and 16, and based on the procedure outlined in Section 2.3, the potential for energy dissipation through cyclic straining of the Fe-SMA is compared to that of S355J2+N and S690QL steels by comparing (a) the stress-strain relations and (b) an equivalent cumulative dissipated energy of the three materials at a given loading history, normalized by their elastic strain energy. Figures 15 and 16b suggest that at a given strain amplitude, all three materials have the ability to dissipate energy through yielding/phase-transformation. However, S690QL and Fe-SMAs have higher elastic strain energies compared to S355J2+N steel. One typical use of yield dampers in seismic resistant design would be to cap the developed inertial forces along the height

of a building during an earthquake at a desired level [41]. Therefore, the higher elastic strain energy of the S690QL and Fe-SMA compared to S355J2+N steel implies that the developed inertial forces along the height of a structure equipped with yield dampers may still be fairly high depending on the amount of the elastic strain energy of the yielding segment. High inertial forces lead to appreciable absolute floor acceleration demands during an earthquake that could potentially cause damage to acceleration-sensitive non-structural content [42, 43, 44, 45, 46].

Referring to Figures 16c and 16d, the normalized cumulative energy dissipation as a function of strain excursion (i.e., half load cycle) is compared for all three materials. In all three cases, the S355J2+N steel with a lower yield stress has a higher potential for normalized energy dissipation followed by the Fe-SMA and S690QL regardless of the imposed strain history (i.e., incremental and constant amplitude). This is because the S355J2+N has the lowest elastic strain energy amongst the three materials. This further supports why lower yield stress steels are preferred over higher strength alloys in buckling restrained braces [35]. Referring to Figure 16c, the normalized cumulative dissipated energy exponentially increases while the strain excursion increases for the incremental protocol (LP7). This is due to the fact that the material does not exhibit stress saturation when it is subjected to the incremental protocol. Whereas, during LP5 (see Figure 16d) the rate of energy dissipation seems to decrease after the first excursion since the stress in all three materials stabilizes under constant strain amplitude.

4. Summary and Conclusions

This paper discusses the characterization of an iron-based shape memory alloy (Fe-SMA), Fe-17Mn-5Si-10Cr-4Ni-1(V,C) (ma,-%), under inelastic cyclic tensile/compressive straining. The examined strain-based loading histories are analogous to those that a steel material experiences as part of a dissipative component within a structure under strong ground motion shaking. Besides the typical mechanical properties of the Fe-SMA, key material properties of interest involve the amount of hardening for a given

loading history, the hardening saturation rate, as well as the material dependency to the strain-rate and temperature.

With regards to the Fe-SMA's mechanical properties, the coefficient of variation for the 0.2 % proportional limit stress is around 15 %, thereby indicating an appreciable variability in the material properties between specimens. This is attributed to potential residual stresses due to straightening of the rebar coils prior to machining of the specimens.

The experimental results indicate an asymmetric stress-strain relation between tensile and compressive straining. This is observable fairly well on the constant amplitude cyclic tests. In particular, there are three distinct tangent moduli in the true stress-true strain relation when the specimen is loaded in compression but not in tension. The first one corresponds to the elastic modulus of the specimen, whereas the last two tangent moduli require microstructure analysis to comprehend their physical bases. Interestingly, the same moduli appear to be dependent on the accumulated compound strain.

Additionally, strain rates between 0.03 %/sec and 0.8 %/sec do not significantly influence the Fe-SMA's behavior. The surface temperature of the material within its gauge length also seems to be fairly stable for those strain rate ranges. However, that temperature increases when the imposed strain rate attains values similar to those expected during earthquake loading. For instance, for a given load protocol, an increase of 75 °C in temperature is observed at 8 %/sec, compared to 2 °C at 0.03 %/sec. The difference in surface temperature is attributed to a higher rate of heat generation than the rate of heat dissipation through conduction and convection for tests at higher strain rates. The observed temperature increase at strain rate loading equivalent to those observed in earthquake loading may influence the phase transformation of the Fe-SMA material. While stress-temperature phase diagrams of this material are only available for tensile loading [20], it is recommended that such diagrams should be derived for the above described loading conditions by means of future experimental studies.

The cyclic hardening/softening behavior of the studied Fe-SMA is dependent on

the strain rate and temperature. Particularly, an increase in stress of around 50 MPa due to cyclic hardening is observed in both tension and compression for the slow rate (0.03 %/sec) and the intermediate rate (0.8 %/sec) for the constant 2 % strain amplitude tests. Whereas cyclic softening of approximately 50 MPa can be observed in both tensile and compressive loading excursions for the same strain amplitude test carried out at a high strain rate (8 %/sec).

Direct comparisons with other commonly used structural steels (S355J2+N and S690QL) reveal that the 0.2 % proportional limit stress of the studied Fe-SMA is higher than the one of a S355J2+N steel, and lower than that of a S690QL steel. The strain at fracture of the studied Fe-SMA is also lower than the corresponding values for both the S355J2+N and S690QL steels. Referring to the constant 2 % strain amplitude with a 0.03 %/sec strain rate load protocol (LP5_SR), the amount of cyclic hardening of the Fe-SMA is larger by around 90 % than that of the S690QL steel and around 15 % smaller than that of the S355J2+N steel. Albeit the total energy dissipation per loading excursion (i.e., half-cycle) of the Fe-SMA is nearly the same with that of a S355J2+N steel, the potential of the latter in yield damper applications to reduce the absolute floor acceleration demands along a building's height is still larger than that of high-yield stress alloys (e.g., Fe-SMA and S690QL). The reason is that low yield stress steels do not have appreciable elastic strain energy.

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Table 1: Summary of mechanical properties for the Fe-17Mn-5Si-10Cr-4Ni-1(V,C) (ma,-%) alloy.

a: For the mean and CoV, the results of LP5_MR and LP5_FR are not taken into account.

| LP | Spec. | E [GPa] | $\sigma_{p,0.01\%}$ [MPa] | $\sigma_{p,0.2\%}$ [MPa] | ε_{frac} [%] | σ_u [MPa] |
|----------------------|-------|---------|---------------------------|--------------------------|--------------------------|------------------|
| LP1 | 1 | 163 | 373 | 534 | 54 | 946 |
| LP2 | 1 | 187 | 282 | 472 | 48 | 953 |
| LP3 | 1 | 178 | 263 | 454 | 44 | 952 |
| LP4 | 1 | 189 | 209 | 442 | - | - |
| LP5_SR | 1 | 196 | 239 | 307 | - | - |
| | 2 | 183 | 358 | 519 | - | - |
| LP5_MR | 1 | 172 | 144 | 435 | - | - |
| | 2 | 187 | 125 | 308 | - | - |
| LP5_FR | 1 | 174 | 96 | 258 | - | - |
| | 2 | 176 | 158 | 467 | - | - |
| LP6 | 1 | 178 | 239 | 438 | - | - |
| LP7 | 1 | 193 | 178 | 419 | - | - |
| LP8 | 1 | 188 | 277 | 469 | - | - |
| LP9 | 1 | 187 | 248 | 450 | - | - |
| Mean ^a | | 184 | 267 | 450 | 49 | 950 |
| CoV [%] ^a | | 5.11 | 22.7 | 13.7 | 10.3 | 0.4 |

Table 2: Comparison of the mechanical properties for the Fe-17Mn-5Si-10Cr-4Ni-1(V,C) (ma,-%) alloy from the present study with reported values.

| Reference | E [GPa] | $\sigma_{p,0.01\%}$ [MPa] | $\sigma_{p,0.2\%}$ [MPa] | ε_{frac} [%] | σ_u [MPa] |
|----------------------|-----------|---------------------------|--------------------------|--------------------------|------------------|
| Koster et al. [21] | 200 | 190 | 310 | - | 993 |
| Ghafoori et al. [22] | 173 | 230 | 546 | 55 | 1015 |
| Hosseini et al. [23] | 173 | 230 | 546 | 55 | 1015 |
| Present Study | 184 | 267 | 450 | 54 | 950 |

Table 3: Fe-17Mn-5Si-10Cr-4Ni-1(V,C) (ma,-%) Fe-SMA and structural steels properties.

a: steel data from [37] b: data from [38] c: data from [39].

| Material | E [GPa] | $\sigma_{p,0.01\%}$ [MPa] | $\sigma_{p,0.2\%}$ [MPa] | ε_{frac} [%] | h [%] | σ_u [MPa] |
|-----------------------|-----------|---------------------------|--------------------------|--------------------------|---------|------------------|
| S355J2+N ^a | 200 | 375 | 362 | 27 ^b | 13.3 | 553 |
| S690QL ^a | 200 | 614 | 714 | 15.5 ^c | 4.37 | 1000 |
| Fe-SMA | 184 | 267 | 450 | 54 | 11.7 | 950 |

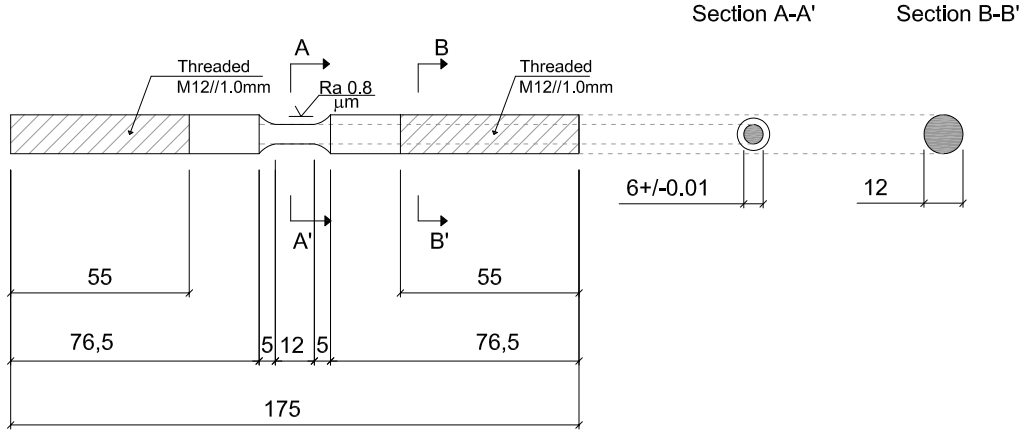


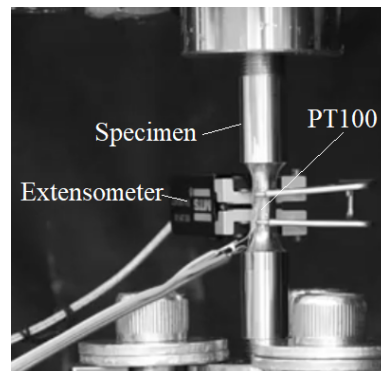
Figure 1: Geometry of the round-bar coupon specimens [mm]



Figure 2: Specimen at 5 % inelastic uniaxial compressive strain

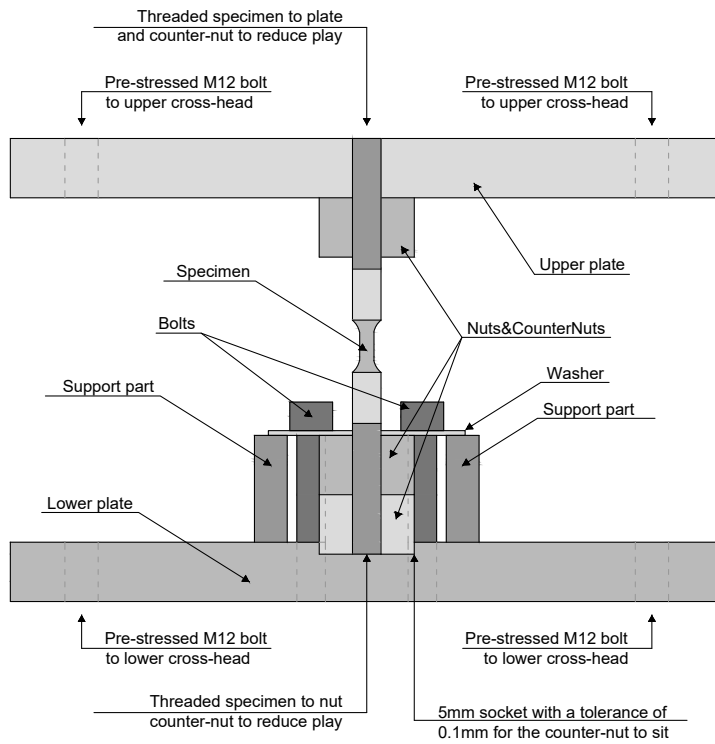


(a) Experimental setup

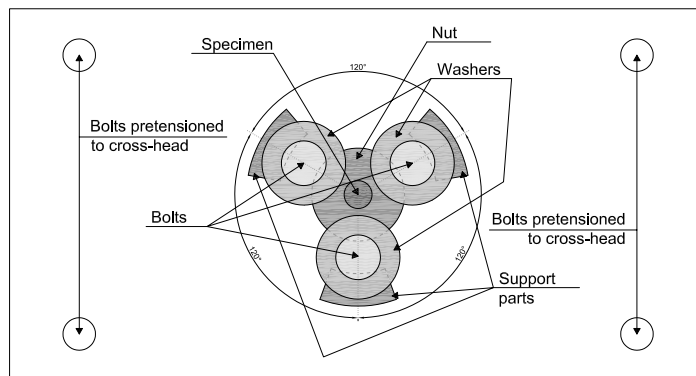


(b) Detailed view of the specimen

Figure 3: Test apparatus and detailed view of round bar coupon specimens

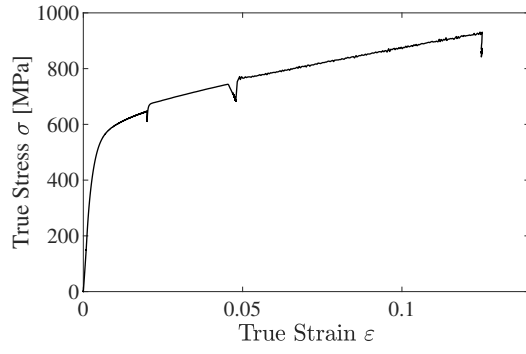


(a) Cross-section

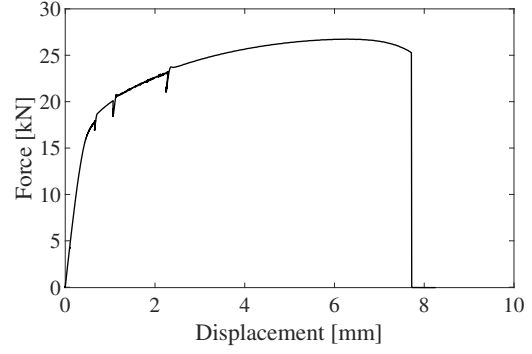


(b) Plan view

Figure 4: Cross-section and plan view of the uniaxial cyclic test setup

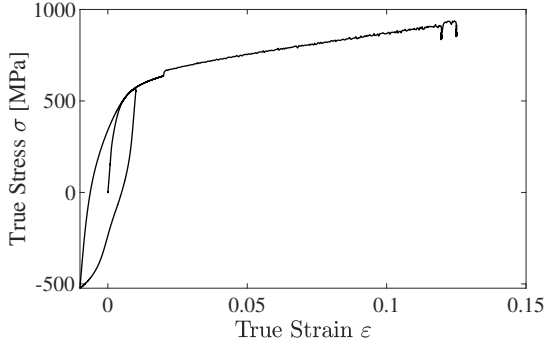


(a) True stress-true strain

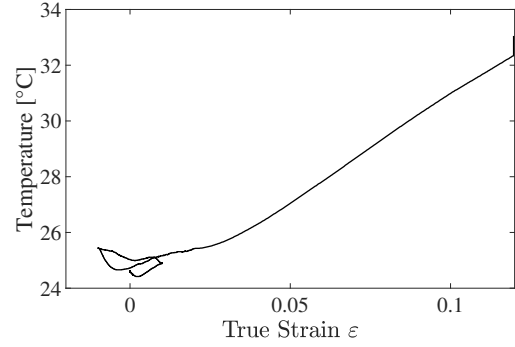


(b) Force-displacement

Figure 5: Fe-SMA material behavior LP1 (0.03 %/sec up to 2 % strain amplitude and 0.8 %/sec after)

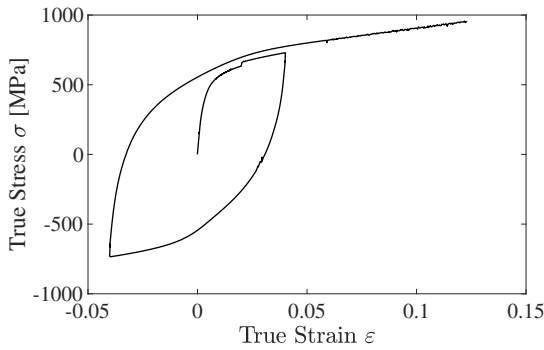


(a) True stress-true strain

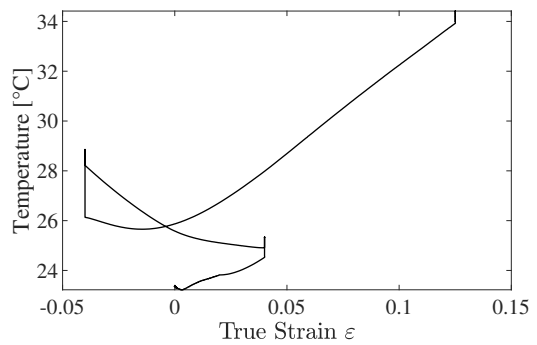


(b) Temperature-true strain

Figure 6: Fe-SMA material behavior LP2 (0.03 %/sec up to 2 % strain amplitude and 0.8 %/sec after)

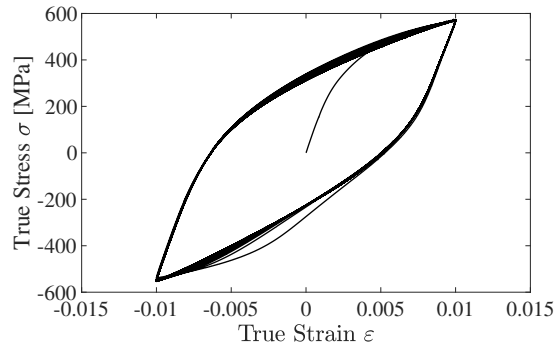


(a) True stress-true strain

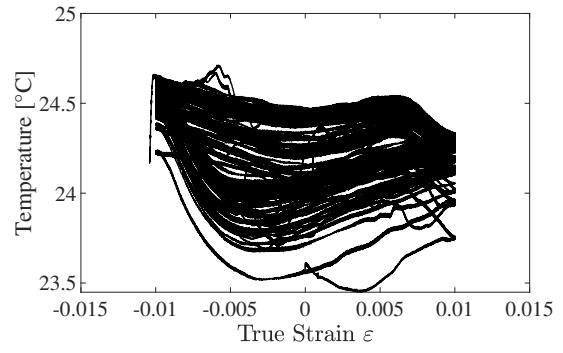


(b) Temperature-true strain

Figure 7: Fe-SMA material behavior LP3 (0.03 %/sec up to 2 % strain amplitude and 0.8 %/sec after)

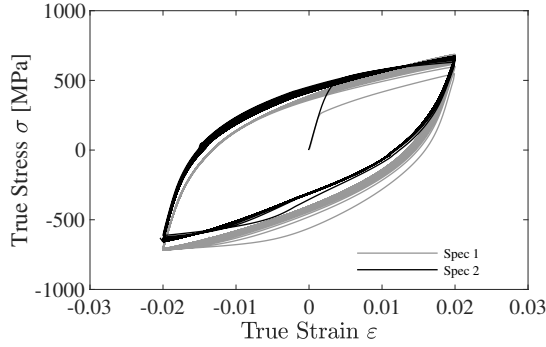


(a) True stress-true strain

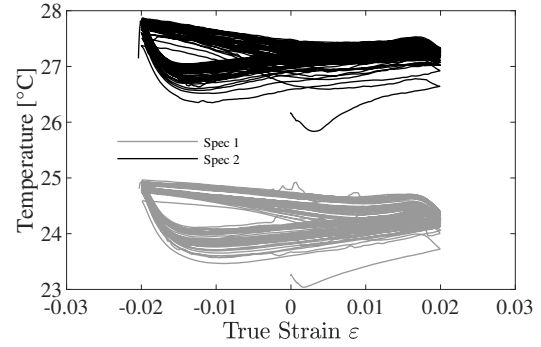


(b) Temperature-true strain

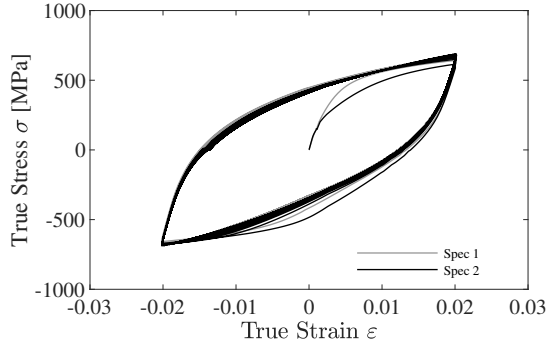
Figure 8: Fe-SMA material behavior LP4 (0.03 %/sec)



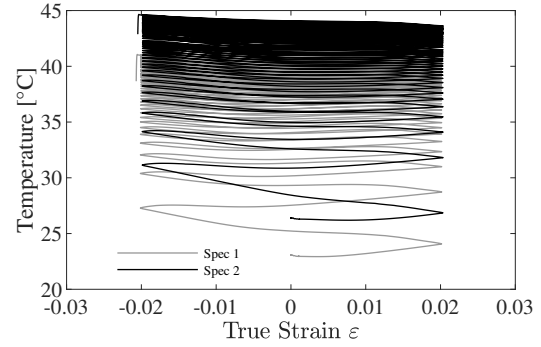
(a) True tress-true strain LP5_SR



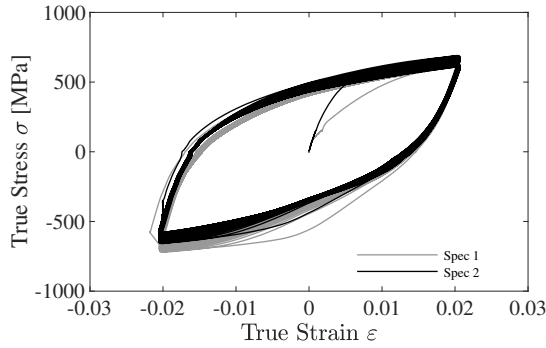
(b) Temperature-true strain LP5_SR



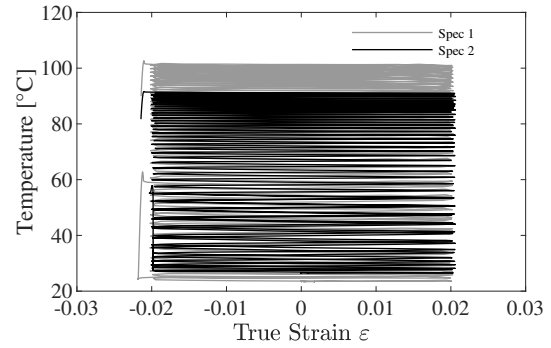
(c) True tress-true strain LP5_MR



(d) Temperature-true strain LP5_MR



(e) True tress-true strain LP5_FR



(f) Temperature-true strain LP5_FR

Figure 9: Fe-SMA material behavior LP5 - influence of strain rate on the true-stress-true strain and the temperature-true strain relationships

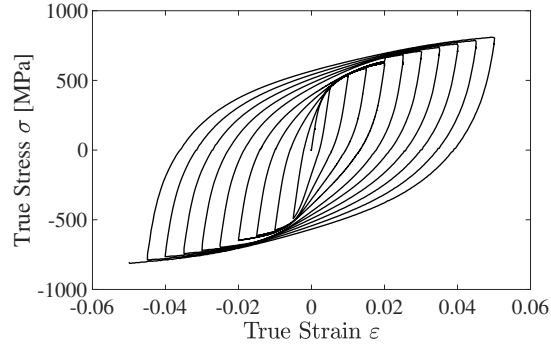
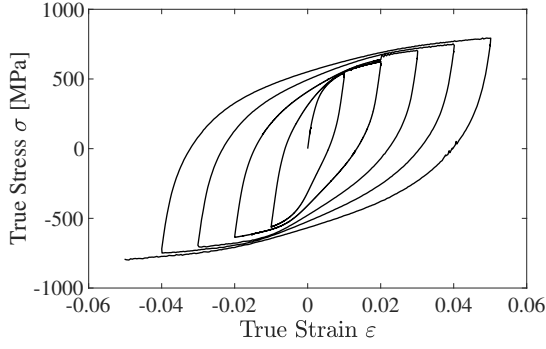
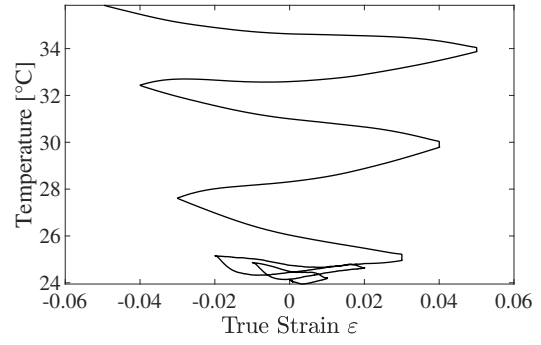


Figure 10: True stress-true strain LP6 (0.03 %/sec up to 2 % strain amplitude and 0.8 %/sec after)

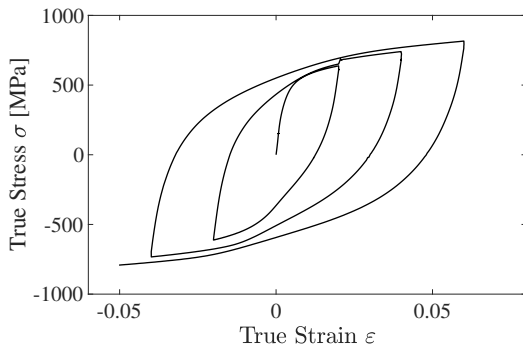


(a) True stress-true strain

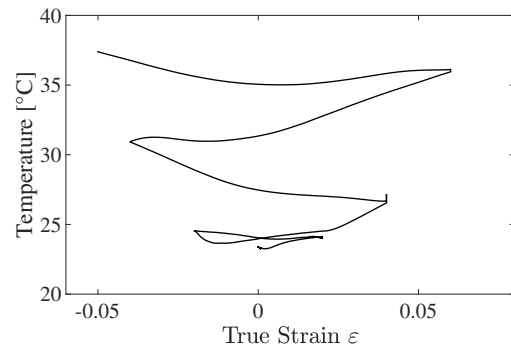


(b) Temperature-true strain

Figure 11: Fe-SMA material behavior LP7 (0.03 %/sec up to 2 % strain amplitude and 0.8 %/sec after)

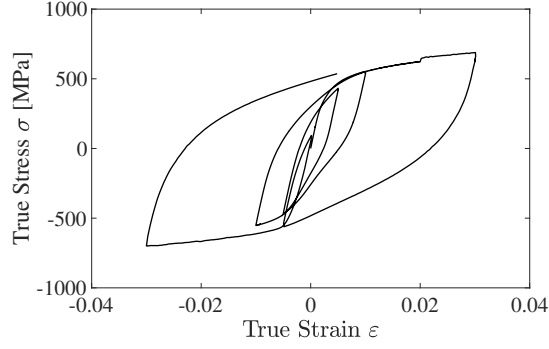


(a) True stress-true strain

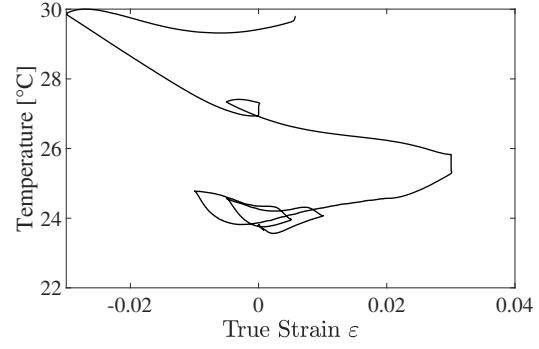


(b) Temperature-true strain

Figure 12: Fe-SMA material behavior LP8 (0.03 %/sec up to 2 % strain amplitude and 0.8 %/sec after)

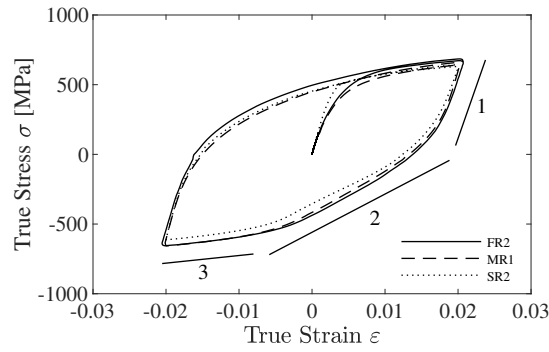


(a) True stress-true strain

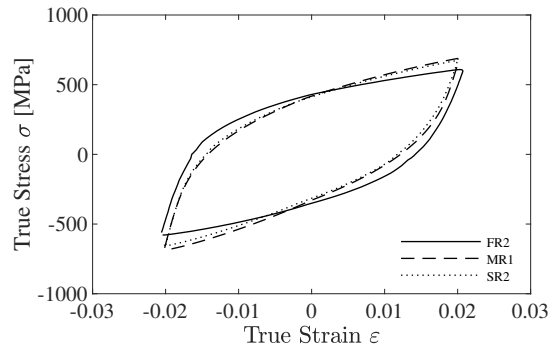


(b) Temperature-true strain

Figure 13: Fe-SMA material behavior LP9 (0.03 %/sec up to 2 % strain amplitude and 0.8 %/sec after)

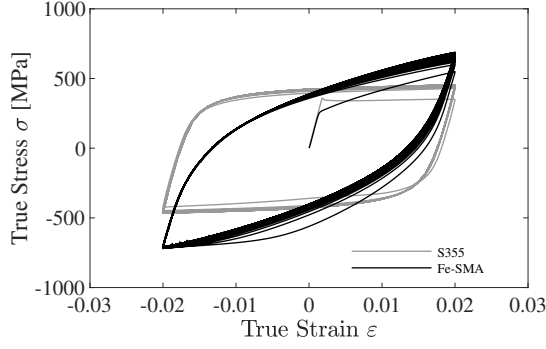


(a) First cycle

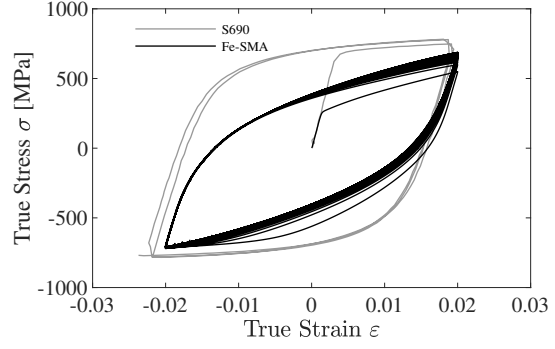


(b) Last cycle (50th cycle)

Figure 14: Comparison of the Fe-17Mn-5Si-10Cr-4Ni-1(V,C) (ma,-%) Fe-SMA true stress-true strain relationship for LP5 at different strain-rates

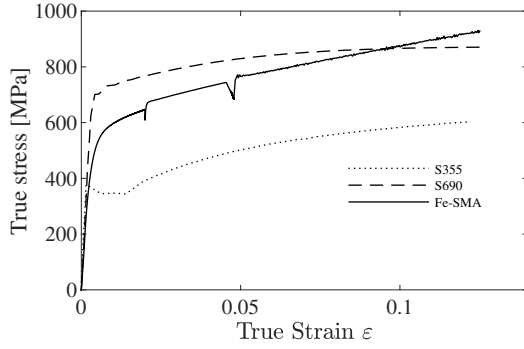


(a) S355J2+N and Fe-SMA

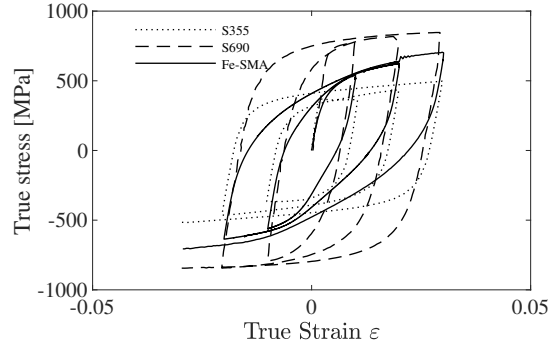


(b) S690QL and Fe-SMA

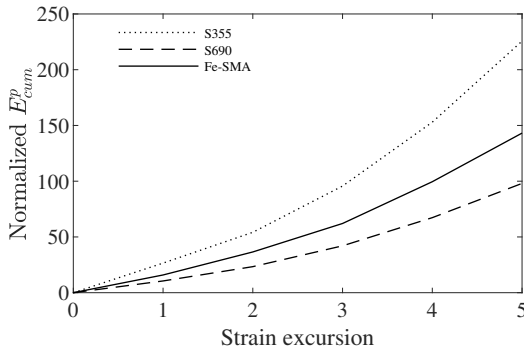
Figure 15: Comparison of the Fe-17Mn-5Si-10Cr-4Ni-1(V,C) (ma,-%) Fe-SMA and structural steels for LP5



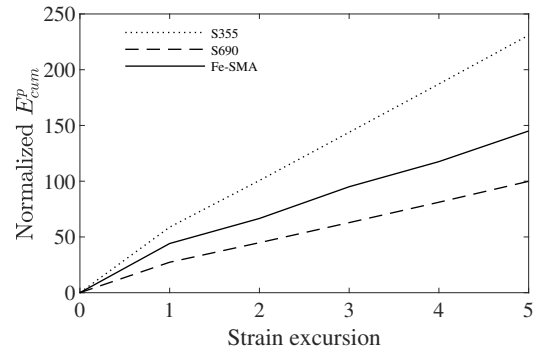
(a) True stress-true strain LP1



(b) True stress-true strain LP7



(c) Normalized energy dissipated LP7



(d) Normalized energy dissipated LP5

Figure 16: Stress-strain and energy dissipation of the Fe-17Mn-5Si-10Cr-4Ni-1(V,C) (ma,-%) Fe-SMA and structural steels for LP1 and LP7